# NUMERICAL ANALYSIS OF THE INTERACTION AMONG FLOWS AT LOW REYNOLDS NUMBERS AND ELASTICALLY MOUNTED CYLINDERS 

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Abstract. The vortex-induced vibration (VIV) phenomenon has drawn the attention of researchers in Engineering for several decades. An example is the riser used for petroleum exploration, in which it is subjected to marine flows that may cause oscillations due to vortex shedding. In this paper, numerical analyses of the phenomena that occur in the interaction among flows at low Reynolds number and elastically mounted cylinders are presented. The simulation is carried out by using the numerical model that uses a semi-implicit two-step Taylor-Galerkin method to discretize the Navier-Stokes equations and the arbitrary lagrangean-eulerian formulation to follow the cylinder movement. The rigid body motion description is calculated by using the Newmark method. Firstly, the characteristics of the vortex generation process for the fixed cylinder are analyzed. In this case, the Strouhal number, the mean drag and the RMS lift coefficients for Reynolds numbers ranging from 90 to 140 are shown. Afterwards, an analysis of a flexible supported cylinder (with a spring and a damper) in transverse direction subject to flows with Reynolds numbers ranging from 90 to 140 is carried out. The cylinder displacement and the vibration frequencies are studied; the synchronization between the vortex shedding and the vibration frequency (lock-in) is analyzed. Similar results to the experimental ones developed by Anagnostopoulos and Bearman (1992) were obtained in this study.

Keywords: Finite element method, Oscillating cylinder, Fluid-structure interaction.

## 1. INTRODUCTION

Vortex-induced vibration (VIV) is a phenomenon that is found in several engineering fields. Some examples are the following: wind can cause oscillations on bridges, slender buildings, chimneys and energy transmission cables; flows with high velocities can induce orbital movements in internal tubes of a heat exchanger; and currents and waves can cause vibration on pipelines.

The wake around a circular cylinder due to a uniform flow leads to variety of complex phenomena, thus causing instability that increases close to the wake. This case has been studied for several decades, and, nowadays, the behavior of the flow is known. For Reynolds number (Re) up to 49, two stationary recirculation zones attached to the cylinder wall are observed. From 49 to 190, the wake is still laminar and it is composed by two periodic staggered rows of alternating vortices (von Kármán vortex shedding). For higher Reynolds numbers (from 190 to 260), the wake becomes tridimensional and progressively turbulent. This regime is followed by a shear layer transition (up to 1200), in which separating shear layers become unstable, and, finally, by the boundary layer transition (around $10^{5}$ ) associated with fast decrease of the drag coefficient. For these regimes, the flow exhibits a periodicity which is known as Strouhal frequency. When a periodic vortex street is well established, this frequency corresponds to that of the vortex shedding frequency; in other cases, in which the von Kármán streets are not clearly visible, the frequency can be defined as the one of the fluctuations of the streamwise velocity component, for example (Placzek et al., 2009).

In many applications, the cylinder oscillates and interacts with the vortex shedding process. For forced oscillations in a range of frequency and amplitude, the cylinder motion is able to control the instability mechanism generated by vortex shedding. One of the most interesting characteristics of this fluid-structure interaction is the synchronization (lock-in) between the vortex shedding and the vibration frequency. Similar phenomena are observed for VIV, in which the flow causes the oscillation of the cylinder at its natural frequency. This frequency depends on the mass, the rigidity and the damping of the cylinder. In this phenomenon, which occurs in a range of flow velocity, the amplitude reaches a peak.

This complex fluid-structure interaction phenomenon is still a good test case to validate the numerical models. Several numerical analyses can be found in the literature for a large range of Reynolds numbers, including Reynolds Averaged Navier-Stokes (RANS) methods (Saghafian et al., 2003; Guilmineau and Queutey, 2004), Large Eddy Simulations (LES) (Breuer, 2000; Pasquetti, 2005; Al-Jamal and Dalton, 2004), Direct Numerical Simulations (DNS) and methods that use finite volume or finite element approximations to solve the Navier-Stokes equations (Anagnostopoulos and Bearman, 1992; Nobari and Naredan, 2006; Mittal and Kumar, 2001).

This paper describes simulations which are carried out by using the numerical model that uses a semi-implicit twostep Taylor-Galerkin method to discretize the Navier-Stokes equations and the arbitrary lagrangean-eulerian
formulation to follow the cylinder movement. The rigid body motion description is calculated by using the Newmark method. Firstly, the characteristics of the vortex generation process for the fixed cylinder are analyzed. In this case, the Strouhal number, the mean drag and the RMS lift coefficients for Reynolds numbers ranging from 90 to 140 are shown. Afterwards, an analysis of a flexible supported cylinder (with a spring and a damper) in transverse direction subject to flows with Reynolds numbers ranging from 90 to 140 is carried out. The cylinder displacement and the vibration frequencies are studied; the synchronization between the vortex shedding and the vibration frequency (lock-in) is analyzed. Similar results to the experimental ones developed by Anagnostopoulos and Bearman (1992) were obtained in this study.

## 2. NUMERICAL MODEL

The numerical model is based on a partitioned scheme, in which the fluid flow and the structure are solved in twoway interaction. Basically, the fluid-structure interaction adopted by the code consists in the following steps: (a) update the variables of the flow from instant $t$ to $t+\Delta t$; (b) impose pressure and viscous stress as a load to the structure; (c) update the variables of the structure from instant $t$ to $t+\Delta t$; (d) impose the body movement to the flow in terms of the updated velocity vector and boundary position.

Basically, updating the variables of the flow consists of following steps (Teixeira, 2001):
a) Calculate non-corrected velocity at $\mathrm{t}+\Delta \mathrm{t} / 2$, where the pressure term is at $t$ instant, according to Eq. (1).

$$
\begin{equation*}
\tilde{U}_{i}^{n+1 / 2}=U_{i}^{n}-\frac{\Delta t}{2}\left(\frac{\partial f_{i j}^{n}}{\partial x_{j}}-\frac{\partial \tau_{i j}^{n}}{\partial x_{j}}+\frac{\partial p^{n}}{\partial x_{i}}-\rho g_{i}-w_{j}^{n} \frac{\partial U_{i}^{n}}{\partial x_{i}}\right) \tag{1}
\end{equation*}
$$

where $\rho$ is the specific mass, $p$ is the pressure, $g_{i}$ are the gravity acceleration components, $\mathrm{v}_{i}$ are the velocity components, $\quad \mathrm{w}_{i}$ are the velocity components of the reference system and $\tau_{i j}$ is the viscous stress tensor $U_{i}=\rho_{\mathrm{V}_{i}}$, $f_{i j}=\mathrm{v}_{j}\left(\rho_{\mathrm{v}_{i}}\right)=\mathrm{v}_{j} U_{i}(i, j=1,2)$.
b) Update the pressure $p$ at $t+\Delta t$, given by the Poisson equation:

$$
\begin{equation*}
\frac{1}{c^{2}} \Delta p=-\Delta t\left[\frac{\partial \tilde{U}_{i}^{n+1 / 2}}{\partial x_{i}}-\frac{\Delta t}{4} \frac{\partial}{\partial x_{i}} \frac{\partial \Delta p}{\partial x_{i}}\right] \tag{2}
\end{equation*}
$$

where $\Delta p=p^{n+1}-p^{n}$ and $i=1,2$.
c) Correct the velocity at $t+\Delta t / 2$, adding the pressure variation term from $t$ to $t+\Delta t / 2$, according to the equation:

$$
\begin{equation*}
U_{i}^{n+1 / 2}=\tilde{U}_{i}^{n+1 / 2}-\frac{\Delta t}{4} \frac{\partial \Delta p}{\partial x_{i}} \tag{3}
\end{equation*}
$$

d) Calculate the velocity at $t+\Delta t$ using variables updated in the previous steps as follows:

$$
\begin{equation*}
U_{i}^{n+1}=U_{i}^{n}-\Delta t\left(\frac{\partial f_{i j}^{n+1 / 2}}{\partial x_{j}}-\frac{\partial \tau_{i j}^{n+1 / 2}}{\partial x_{j}}+\frac{\partial p^{n+1 / 2}}{\partial x_{i}}-w_{j}^{n+1 / 2} \frac{\partial U_{i}^{n+1 / 2}}{\partial x_{i}}-\rho g_{i}\right) \tag{4}
\end{equation*}
$$

The classical Galerkin weighted residual method is applied to the space discretization of Eq. (1), (2), (3) and (4), and a triangular element is employed. In the variables at $t+\Delta t / 2$ instant, a constant shape function is used, and in the variables at $t$ and $t+\Delta t$, a linear shape function is employed (Teixeira and Awruch, 2001). The mesh velocity vertical component $w_{2}$ is computed to diminish element distortions, keeping prescribed velocities on moving and stationary boundary surfaces. The mesh movement algorithm adopted in this paper uses a smoothing procedure for the velocities based on these boundary lines. The updating of the mesh velocity at node $i$ of the finite element domain is based on the mesh velocity of the nodes j that belong to the boundary lines.

In order to update the rigid-body motion structure, it is necessary to calculate displacements and rotations of a hypothetical concentrated mass at its gravity center. In this study case, there is only movement in transverse direction (one degree of freedom - DOF) and, consequently, displacement, velocity and acceleration in this direction are the variables to be determined at each time step. To update the variables of the structure, the rigid movement of the cylinder is calculated at each instant, after the variables of the flow (pressure and viscous stress) are known. For this study case, one DOF dynamic equation is considered for the transverse direction, as follows:

$$
\begin{equation*}
m \ddot{y}+c \dot{y}+k y=F \tag{5}
\end{equation*}
$$

where $\ddot{y}, \dot{y}$ and $y$ are the transverse acceleration, velocity and displacement, respectively; $m$ is the mass; $c$ is the damping coefficient; $k$ is the stiffness; and $F$ is the dynamic force. In this code, Eq. (5) is discretized in time by using the implicit Newmark method (Bathe, 1996) and the acceleration, the velocity and the displacement in transverse direction are calculated at each time step.

## 3. NUMERICAL SIMULATIONS

The case study consists of a cylinder (diameter and mass equal to 0.0016 m and 0.2979 kg , respectively) subject to a uniform water flow (specific mass, $\rho$, and viscosity, $\mu$, equal to $1000.0 \mathrm{~kg} / \mathrm{m}^{3}$ and $0.001 \mathrm{~kg} /(\mathrm{ms})$, respectively). The cylinder is mounted on a spring and a damper in transverse direction and fixed towards the flow. The spring stiffness, $k$, is equal to $579 \mathrm{~N} / \mathrm{m}$ and the damping coefficient, $c$, is equal to $0.0325 \mathrm{~kg} / \mathrm{s}$. The natural frequency of this system is $f n=7.016 \mathrm{~Hz}$.

The influence of the size of the computational domain was analyzed for Reynolds number equal to 135 . The best computational domain that satisfied both accuracy and computational cost criteria was a rectangle 0.320 m wide and 0.384 m long, as shown in Fig. 1. The cylinder center is located at the center of the domain in transverse direction to the flow and 0.160 m from its left side in longitudinal direction. After analyzing the mesh convergence, a finite element unstructured mesh composed by triangles with 200 element sides around the cylinder was used. Element sizes increase gradually towards the boundaries of the domain. The mesh has 298143 nodes and 595526 elements. A constant velocity is imposed on the left side; on the superior and the inferior walls, a slide condition is imposed; and the right side is free, but null pressure is imposed on its middle. The time step used for the simulations and 5.0 $\left(10^{-5}\right) \mathrm{s}$.


Figure 1. Numerical domain.
First, the behavior of the flow considering the fixed cylinder is analyzed. Specifically, the flow characteristics at $\operatorname{Re}=105$ (velocity $U$ equal to $0.065625 \mathrm{~m} / \mathrm{s}$ ) are studied. Figure 2 shows the velocity vectors at eight instants along one period of vortex formation, respectively. The vortex formation was clearly observed, showing two different regions behind the cylinder where the flow separation occurs. Near the cylinder surface, while the larger vortex is in one direction, the opposing vortex is in another one.
$\operatorname{Drag}\left(\mathrm{F}_{\mathrm{D}}\right)$ and lift $\left(\mathrm{F}_{\mathrm{L}}\right)$ forces on the cylinder obtained by numerical simulation for $\mathrm{Re}=105$ are shown in Fig. 3. The drag force has a periodic behavior with a little variation around 0.0045 N , whereas the lift force has a periodic behavior with amplitude equal to 0.0012 N and frequency equal to 6.828 Hz .

Figure 4 shows a comparison among the Strouhal numbers obtained by numerical results (where $f$ is equal to the lift force frequency) for Re from 90 to 140 and Willianson's results (Willianson, 1989), which are based on experimental data. The numerically obtained Strouhal numbers have good agreement with experimental ones; the mean difference was only $0.03 \%$. The mean drag coefficients $\left(\mathrm{C}_{\mathrm{D}}\right)$ and the Root mean square (RMS) lift coefficients $\left(\mathrm{C}_{\mathrm{L}}\right)$ for the same range of Re are shown in Fig. 5 and 6; these values are compared with those obtained by Poldsziech and Grundmann (2007) and Baranyi and Lewis (2006), respectively. The former uses a Spectral Element method and the latter uses a Grid based method. The mean differences among results obtained by this study and by those authors' were $0.8 \%$ and $0.2 \%$ regarding $C_{D}$ and $C_{L}$, respectively. It is worth mentioning that the high accuracy of the parameters $S t$ and $C_{L}$ shows the capacity of the model to reproduce the frequency and the magnitude of the force that is imposed over the structure in the fluid structure interaction process.


Figure 2. Velocity vectors at eight instants along a period of vortex formation.

(a)

Figure 3. Drag (a) and lift (b) forces for fixed cylinder with $\mathrm{Re}=130$.


Figure 4. Strouhal numbers in relation to Reynolds numbers.


Figure 5. Mean drag coefficient $\left(\mathrm{C}_{\mathrm{D}}\right)$ in relation to Reynolds numbers.


Figure 6. RMS lift coefficient $\left(C_{L}\right)$ in relation to Reynolds numbers.
Afterwards, the interaction among a cylinder mounted on an elastic fixing in transverse direction and flows with several Reynolds numbers, from 90 to 140, is analyzed. Figure 7 shows the relation between the amplitude (Y) of the cylinder oscillation and its diameter (D) in the function of Reynolds numbers ( 90 to 140). Figure 8 shows the relation between the frequency of vibration and the natural frequency $(f / f n)$ in the function of Reynolds numbers.

The numerical results show that the lock-in phenomenon was captured for Reynolds numbers between 105 and 110. This is observed due to the increase of the amplitude and the equality of vibration and natural frequencies. Out of the lock-in region, the amplitudes are negligible and the vibration frequencies are away from the natural frequency.

The differences between experimental (Anagnostopoulos and Bearman, 1992) and numerical amplitudes in the Reynolds number range in the lock-in region were observed by Dettmer and Peric (2006). These authors used a model that employs the stabilized low order velocity-pressure finite elements, an arbitrary Lagrangian-Eulerian formulation and the discrete implicit generalised- $\alpha$ method for the rigid body motion. According to the authors, these differences can be explained comparing the numerical domain and boundary conditions and the real situation of the experience which was done in a 0.70 m deep channel where 0.12 m of the cylinder was submerged. The lack of a horizontal plate in the end of the submerged cylinder allowed the vortex shedding in this region. This fact and the influence of the free
surface contribute to develop the tridimensional behavior of the flow, unlike the numerical conditions.


Figure 7. Amplitude (Y/D) in relation to Reynolds numbers.


Figure 8. Frequency of vibration $(f / f n)$ in relation to Reynolds numbers.
Figure 9 shows the behavior of the drag and the lift forces for oscillating and fixed cylinders with $\mathrm{Re}=105$. Unlike the drag force for the fixed cylinder, which has a small harmonic variation around 0.0045 N , this force oscillates between 0.0050 N and 0.0073 N at a frequency equal to 14.085 Hz (almost twofold the frequency in transverse direction) for the oscillating cylinder. In the transverse direction, the lift forces differ in terms of frequency and amplitude. The amplitudes for the fixed and the oscillating cylinders are 0.0012 N and 0.0014 N and their frequencies are 6.828 Hz and 6.984 Hz , respectively. Both frequencies are related to the vortex shedding and the latter is closer to the natural frequency of the dynamic system of the cylinder ( $f n=7.016 \mathrm{~Hz}$ ).

A representative case out of the lock-in region ( $\mathrm{Re}=123$ ) was chosen to show its different force behavior. Figure 10 shows the drag and the lift forces along the time for $\mathrm{Re}=123$. The drag forces have little variation around 0.0062 N for both fixed and oscillating cylinders, with more disturbances in the case of the fixed cylinder. The drag force for the fixed cylinder has a harmonic behavior with amplitude and frequency equal to 0.0020 N and 8.313 Hz , respectively. For the oscillating cylinder, this force oscillates periodically (frequency of 8.163 Hz ) within an envelop with minimum and maximum amplitudes of 0.0014 N and 0.0020 N , respectively. Although these amplitudes are higher than those in the previous case $(\mathrm{Re}=105)$, the displacements are smaller, because this case $(\mathrm{Re}=123)$ is out of the lock-in region.


(b)

Figure 9. Drag (a) and lift (b) forces for cylinder with $\mathrm{Re}=105$.



Figure 10. Drag (a) and lift (b) forces for cylinder with $\operatorname{Re}=123$.

## 4. CONCLUSIONS

In this paper, numerical analyses of the phenomena that occur in the interaction among flows at low Reynolds numbers and elastically mounted cylinders were presented. The simulation was carried out by a numerical model that uses a semi-implicit two-step Taylor-Galerkin method to discretize the Navier-Stokes equations and the arbitrary lagrangean-eulerian formulation to follow the cylinder movement. The rigid body motion description is calculated by using the Newmark method.

For the fixed cylinder, the behavior of the vortex formation was correctly reproduced: near the cylinder surface, the larger vortex was in one direction, while the opposing vortex was in another one. The Sthrouhal numbers were calculated for Reynolds number range from 95 to 140 . These values were similar to those obtained by Willianson's experiments. The mean drag and the RMS lift coefficients were also calculated and compared with those obtained by Poldsziech and Grundmann (2007) and Baranyi and Lewis (2006), respectively. Minor mean differences ( $0.8 \%$ and $0.2 \%$, respectively) showed the accuracy of the numerical simulation.

The lock-in phenomenon was captured for Reynolds numbers between 105 and 110, characterized by the increase of the amplitude and the equality of vibration and natural frequencies. Comparing the numerical and the experimental results (Anagnostopoulos and Bearman, 1992), some differences were observed due to the presence of the tridimensional effects of the experiment that were not considered in this numerical simulation.

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