ABLATION AND MECHANICAL PROPERTIES OF QUARTZ PHENOLIC COMPOSITES

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Abstract. *Quartz phenolic composites have been applied to thermal protection systems(TPS) for reentry vehicles since the late fifties due to their excellent ablation resistance and mechanical performance. The TPS must withstand the aggressive reentry environment, such as atomic oxygen, when submitted to very high temperatures (> 1000° C) and heat flux.*

The ablative performance of composites is influenced by both base materials and environmental parameters during the ablation process.

For TPS systems the phenolic resin is usually used as the base matrix due to its ability to form a stable char during decomposition. This char plays an important role in the absorption of the heat generated during the ablation process. During re-entry parts of the charred matrix can be abrasively removed by shear force due to high pressure and velocity.

In this work the ablation and mechanical properties of quartz phenolic composites were evaluated in order to have the range of properties suitable for the use of these materials as thermal protection systems for space vehicles.

Quartz fabric having an areal weight of 680 g/m^2 and a resole-type phenolic resin were used to prepare the composites. The resin has a viscosity of 165 mPa.s at 20°C. The prepreg material was cured by heating under pressure of 100 bar in a mold. The resin content of the prepreg obtained was about 50%.

The mechanical properties evaluated were, tensile , shear and flexural strength. The results obtained showed that this material has average values of 38,5 MPa, 52 MPa and 85 MPa for tensile, shear and flexural strength, respectively.

The ablative tests were carried out in a high-energy air plasma in ambient atmosphere and the mass losses were measured for different exposure time.

***Keywords**: ablation, quartz phenolic, mechanical properties, thermal protection systems*

1. INTRODUCTION

Thermal protection systems (TPS) are essential for the successful launch and operation of all spacecraft, manned or unmanned. TPS must be good enough to prevent excessive heat from destroying or damaging a vehicle or its contents. Of course the selection of a TPS depends on the mission of the spacecraft.

Also there are different mechanisms of thermal protection. The one investigated in this work is the ablative system . Ablative materials (or ablators) are materials used in Thermal Protection Systems (TPS) that dissipate heat generated by atmospheric friction. Ablative materials are generally employed on non reusable planetary probes.

Ablative materials work by insulating a great amount of heat through a phase change. When the surface of the ablative material (normally a laminate made with resin binder and a fiber compound) reaches a limiting temperature, the resin begins to decompose and absorbs a large part of the heat, preventing it from passing to the backup materials "(Sykes, 1967)". The porous layer that is formed after this degradation ("char") is very important because it acts as an insulator while the material continues to decompose and outgas "(Knop, 1985 and Kanno, 1993)". The char depth and the surface temperature continue to rise until at a certain surface temperature the char will be removed by mechanical shearing, melting and chemical reaction..

For TPS applications, one of the most important requirements is the low thermal conductivity, to prevent an increase of temperature in the back face of the composite, transferring heat to the payload structure. Ablation products injected into the flow field and surface recession alter the flow environment recession. Thus, these processes must be modeled to obtain accurate aero-thermodynamic predictions. "Figure 1" shows schematically the influence of atmosphere, materials properties and aero-thermodynamic loads in the process of ablation.

This paper presents preliminary mechanical properties (tensile, shear and flexural strength) and the ablative properties of a quartz phenolic composite. The ablative properties were obtained in an arc jet plasma that produces jets with gas enthalpies comparable to those encountered during atmospheric reentry.

In order to analyze the microscopic aspects of the samples after exposure to the plasma, the scanning electron microscope (SEM) was utilized.

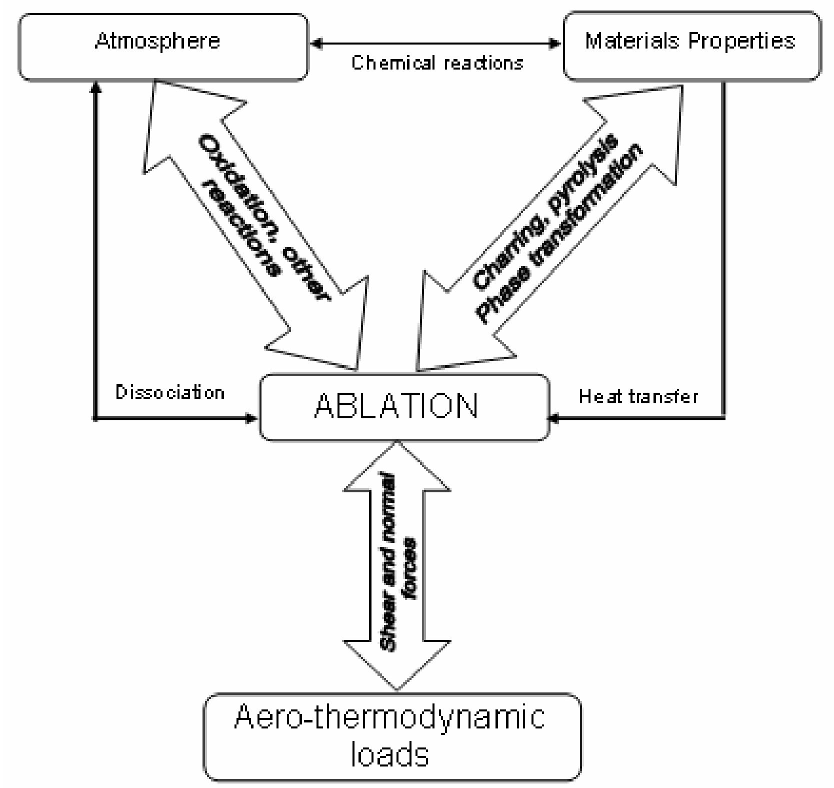


Figure 1. Schematic view of the influence of atmosphere, materials properties and aero-thermodynamic loads in the process of ablation

2. EXPERIMENTAL PROCEDURE

The material analyzed is a phenolic composite containing a bi-directional quartz fabric, having an areal weight of 680 g/m^2. The properties of silica fibers, reported in the manufacture`s literature, are : density of 2.15 g/cm3, tensile strength of 6 GPa , tensile modulus of 78 GPa and elongation to failure of 7.7%. The composite was manufactured by stacking up the fabric by layer and impregnating with a resole-type phenolic resin.

The phenolic resin has a viscosity of 165 mPa.s at 20°C. The impregnation process was carried out in the enterprise Plastiflow Ltd, Curitiba-PR. The prepreg material was cured by heating under pressure of 100 bar in a mold. The resin content of the prepreg obtained was about 50%.

The cure was carried out in a multi-stage cycle in order to increase the wetting of the fibers and the jellification process, up to the final temperature of 187°C. After wetting, the composites presented a volume of fiber of 55-60%.

3. MECHANICAL TESTS

3.1. Tensile test

Tensile test was carried out in an Instron universal testing machine by using the specimen geometry shown in "Fig.2". The tests were carried out by following the ASTM C1275 – "Monotonic Tensile Behavior of Continuous Fiber-Reinforced Advanced Ceramics with Solid Rectangular Cross-Section Test Specimens at Ambient Temperature".

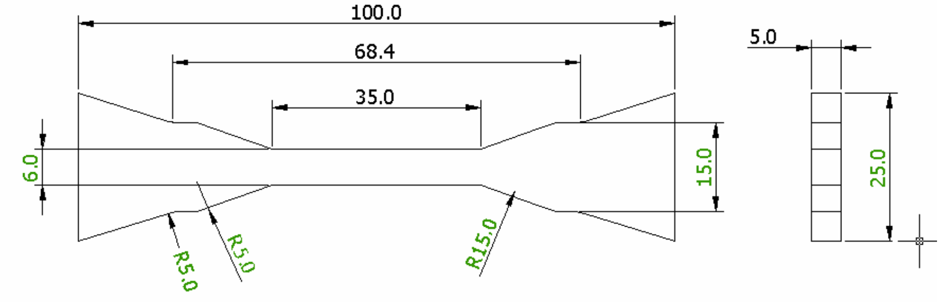


Figure 2. Geometry of test specimen for quartz phenolic composite

3.2. Shear strength test (Iosipescu)

In this work, the shear strength test was carried out in an Instron mechanical testing machine using a test speed of 0,5 mm/min . The test was carried out by following the ASTM D 5379- "Standard Test Method for Shear Properties of Composite Materials by the V-Notched Beam Method".

Although the Iosipescu test was used in the past mainly for metals, since 1992 was defined as the standard method for composite materials.

The fixture of the sample coupon is shown in "Fig. 3". The square area between notches is referred to as the test region. "Figure 4" shows the geometry specimen utilized. The angle of the notches is 90^o. The radius of the notch tips is 1.3 mm. The thickness of specimen was 5.5 mm. Five samples were tested for as-cured and post-cured conditions.

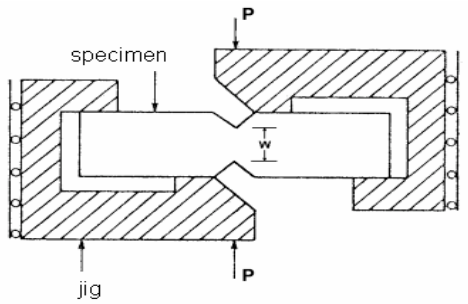


Figure 3. Scheme of the Iosipescu test jig

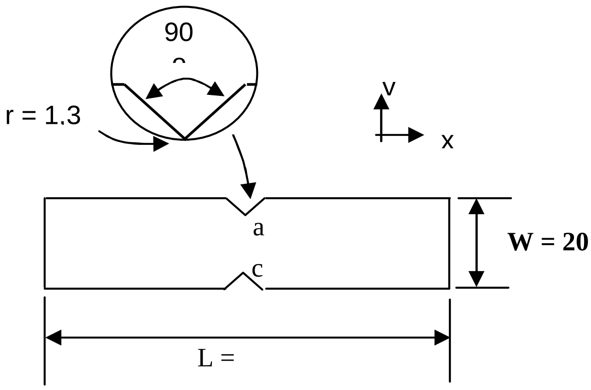


Figure 4. Specimen geometry used in Iosipescu test

3.3. Flexural Test

Flexural test was performed according to ASTM D790 – "Standard Test Method for Flexural Properties of Unreinforced and Reinforced Plastics and Electrical Insulating Materials", in four-point bending mode. The specimens have a width of 10 mm and 2.5 mm thickness. The testing set-up is shown in "Fig. 5". The test speed was 1 mm/min. Ten specimens were tested. Flexural stress and flexural modulus were calculated according to the following equations:

$$Flexure\ stress = \frac{3.load.span}{4.width.(thickness)^2} \qquad Flexure\ modulus\ (E_f) = \frac{0{,}17.span^3.m}{width.depth^3}$$

where "m" is the slope of flexure stress as a function of deflection.

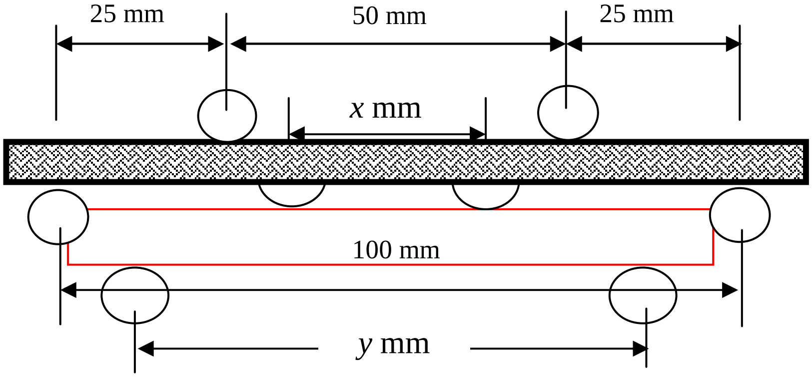


Figure 5. Sample geometry for flexural test

4. ABLATION TEST

A plasma torch test was performed to investigate the ablation property of quartz phenolic composites. The operation was carried out in atmospheric pressure. A DC arc plasma system was designed for continuous working at power up to 50 kW. The intensity of current was adjusted to 135A with tension of 300V. So that, the maximum achievable power obtained, due the power supply, is about 30kW, that gives a plasma enthalpy of about 5,5MJ/kg. The gas flow was maintained at 4.5×10^{-3} kg/s.

The specimen was placed vertically to the flame direction in air. The ablation test was carried out in 10 seconds time and the distance between the nozzle tip of the plasma gun and the front surface of the specimen was varied from 10 to 18 cm. The surface of the samples reaches temperatures in the range of 900 to 1600 °C and it was measured by an optical pyrometer Mod. IR-AH 3SU Chino. "Figure 6" shows the schematic illustration of the apparatus utilized for ablation test using an arc plasma torch.

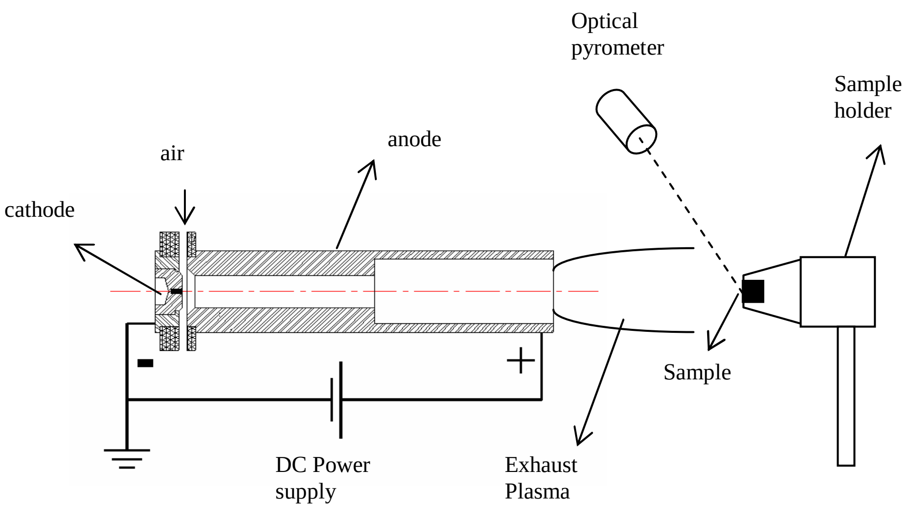


Figure 6. Schematic illustration of the apparatus for ablation test using an arc plasma torch

The burn-through time was measured. The erosion rate was calculated by dividing the specimen thickness or the weight change before and after the test into a burn-through time for each specimen.

One representative sample was used for each condition of testing. The samples were fabricated by Plastiflow Ltd. The specimens were cut in samples with diameter of 1,6 cm and thickness of 1,2 cm.

5. RESULTS AND DISCUSSION

5.1 MECHANICAL TESTS

The tensile test results are shown in “Fig. 7”. The tensile strength of the plain weave quartz phenolic composite was 38 ± 5 MPa. There is a lack of available data in the literature to compare the results obtained in this work, mainly because quartz phenolic composites are used in sensitive areas of aerospace technology. “(Kumara ,2005)” reported 100 MPa for the tensile strength of quartz phenolic composites, which is in the range of the results found in this work. Different sorts of quartz fibers give rise to different composite properties.

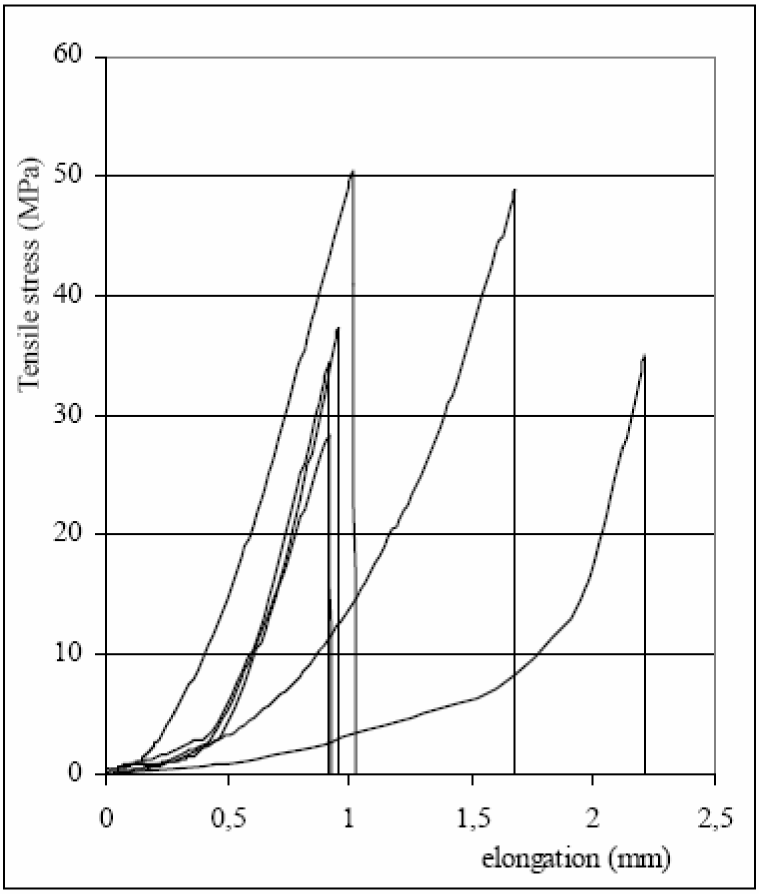


Figure 7. Tensile stress as a function of elongation for plain weave quartz phenolic composite

“Figures 8 and 9” show the results of the Iosipescu test. The quartz phenolic composites were prepared by stacking plain weave fabric plies and the counter-reacting forces were applied perpendicular do the 0/90^o orientation. Considering the axis system for test specimen, the shear modulus to be measured is G_{yx}. The average in-plane Iosipescu shear strength and the shear modulus were measured in as-cured and post-cured samples. The Iosipescu shear strength for the as-cured sample has a value of 19 ± 2 MPa, and is shown in “Fig. 8”. For the post-cured sample, shown in “Fig. 9” , the value is 52 ± 2 MPa. Shear modulus were in the range of 2,5 – 3,5 GPa for both as-cured and post-cured samples.

The as-cured quartz phenolic composite has a higher deformation up to failure (~30000 μm/m), in comparison with the post-cured one (~4500 μm/m). This means that post-cure is beneficial for composite properties providing it is not overcured, which may lead to property degradation. The in-plane shear strength and shear modulus for laminated composites is highly dependent on the matrix properties. The curve shows a typical non-linear behavior up to failure shear stress. For composites having reinforcing fibers perpendicular to the shear loading direction, the failure mode occurs mainly by fiber slipping and debonding at the fiber/matrix interface. On the other hand, in composites having reinforcing fibers parallel to the shear loading direction, the failure may occur by an interlaminar crack at the sample gage length. In any case, shear properties are mainly dominated by the matrix and the fiber/matrix interface and the failure modes are associated with shear deformation mechanism.

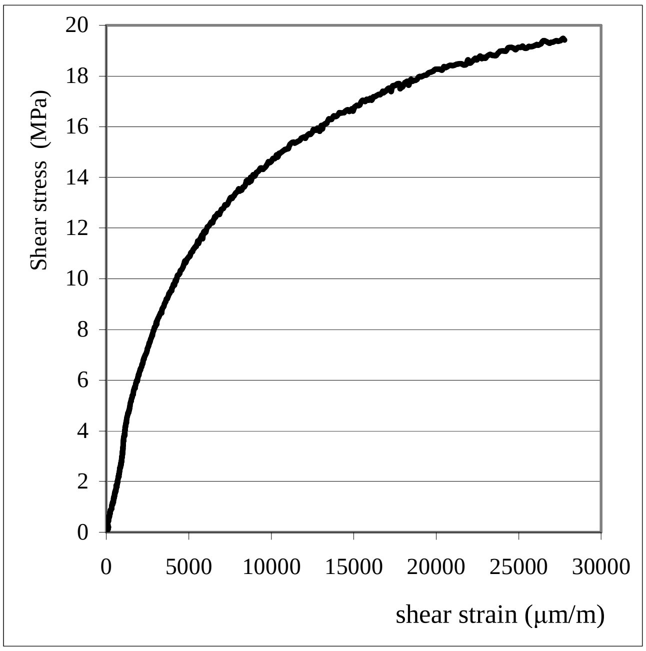


Figure 8. Shear strength for the as-cured samples

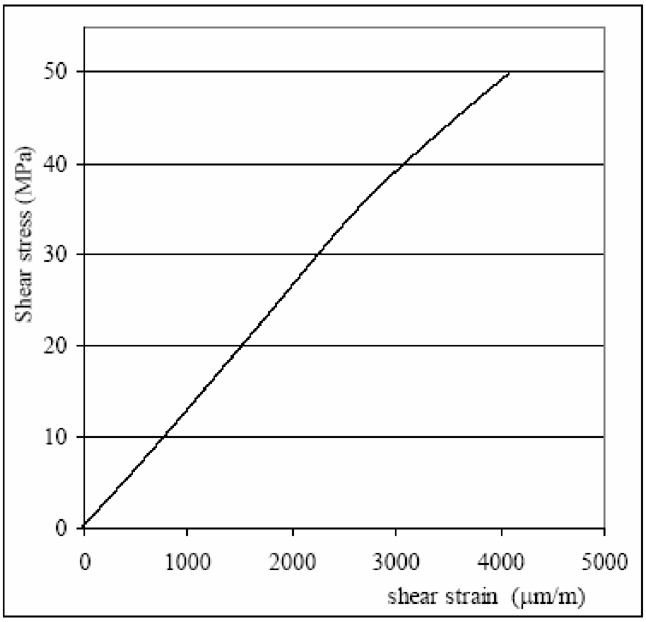


Figure 9. Shear strength for post-cured samples

"Figure 10" shows the results for flexural strength test. The value obtained for quartz phenolic composite was 85 ± 25 MPa. A high scatter was found for the results which may be a result of uneven defects such as fiber misalignment in the composite.

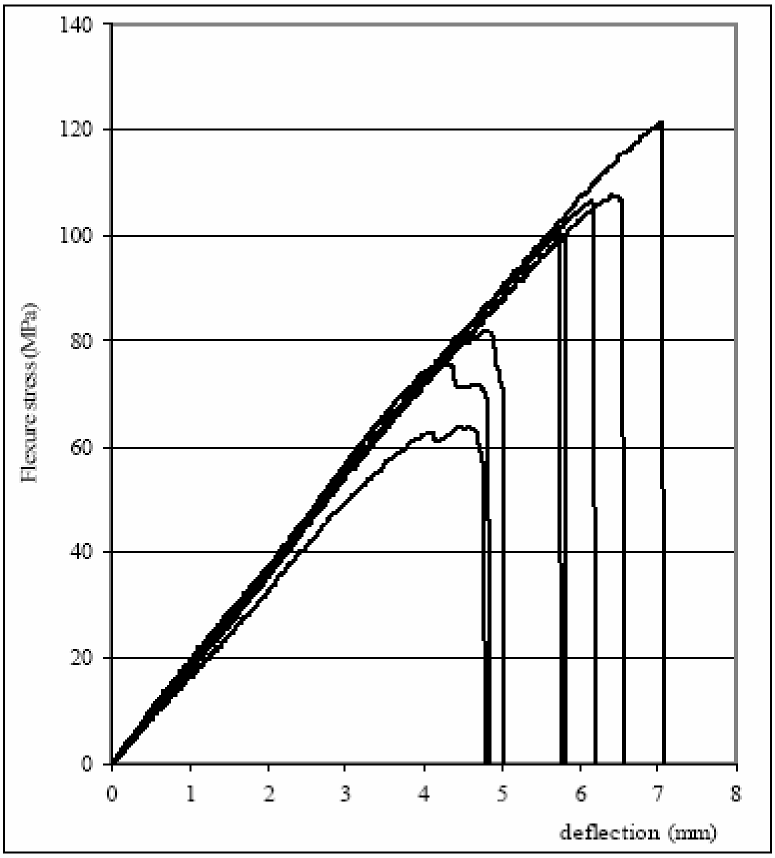
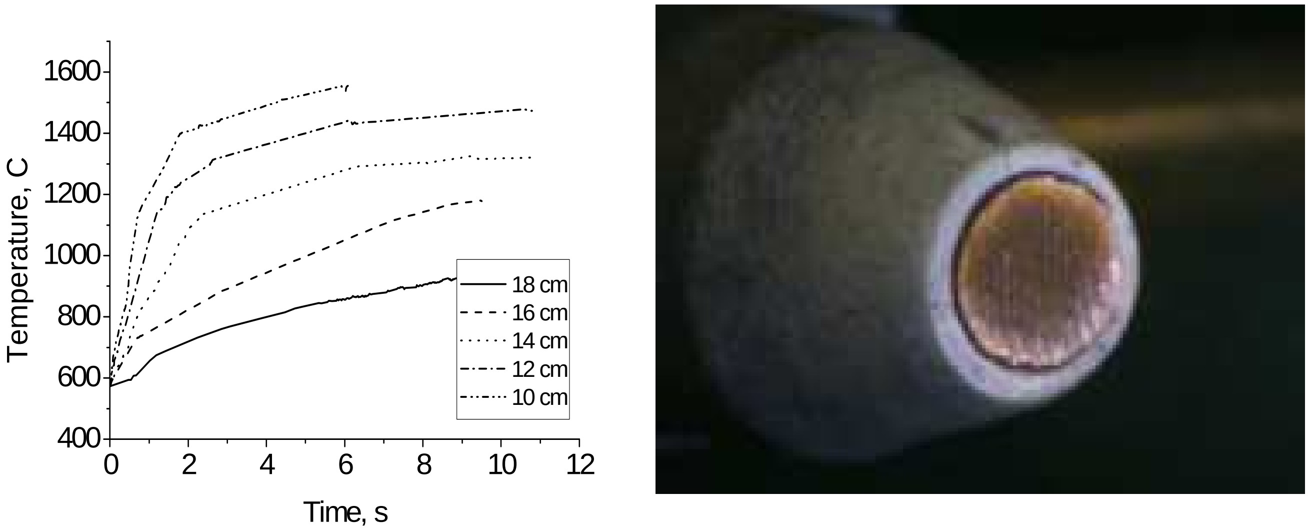


Figure 10. Flexural strength as a function of deflection

5.2. ABLATIVE TEST

"Figure 11 (A)" shows the behavior of the surface temperature with the exposure time for different distances from the plasma torch. The temperature increases during initial phase of heating with subsequent saturation at certain temperature which depends on the value of heat flux density. "Figure 11 (B)" shows a picture of the front surface of the test specimen during testing.

Experimental data with respect to specific mass loss rate of quartz as a function of maximum surface temperature are shown in "Fig. 12".. It was observed that the specific mass loss rate increases exponentially with the maximum surface temperature and varies within a range of about 2.0 10^{-2} kg/m^2s to 8.5 10^{-2} kg/m^2s.



(A) (B)

Figure 11. (A) Surface temperature versus time (B) Front surface of the test specimen

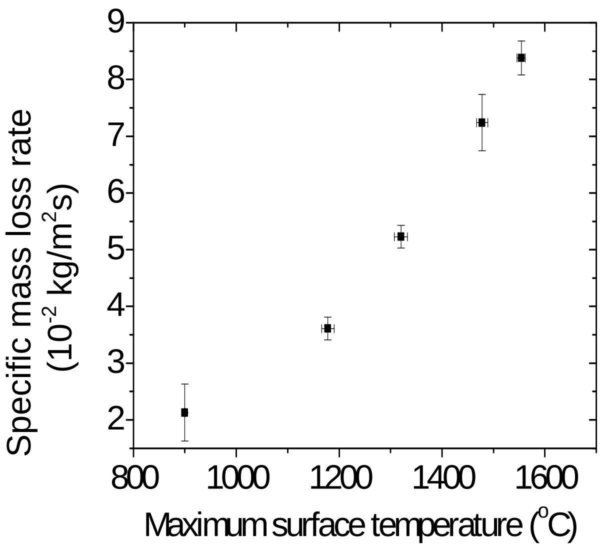


Figure 12. Specific mass loss rate as a function of maximum surface temperature

"Table 1" shows the values of burn depth as a function of the distance between the nozzle tip of the plasma gun and the front surface of the specimen. The thickness of the burned surface increases as the distance decreases. The thickness of the burned surface is also a function of the exposure time of the plasma torch. Although not investigated in this work, the higher the exposure time to the plasma torch the deeper must be the burned surface of the specimen.

"Figure 13" shows a view of a burned surface of the quartz phenolic composite showing the aspect of the surface char generated during the burning testing and the underneath fiber layer. "Figure 14" shows pictures of the burned surfaces of the specimens. The dark areas in the picture are the burned region of the specimens.

As can be seen the burn depth is directly related to the surface temperature and also to the heat flux in the sample. The maximum surface temperatures are shown in Table 1.

Table 1. Burn depth as a function of testing distance for the specimens submitted to the plasma torch testing

Specimen	Testing distance (cm)	Burn depth(mm)	Maximum surface temperature (°C)
A	10	2,5	1600
B	14	1,0	1300
C	16	0,5	1200
D	18	0,2	920

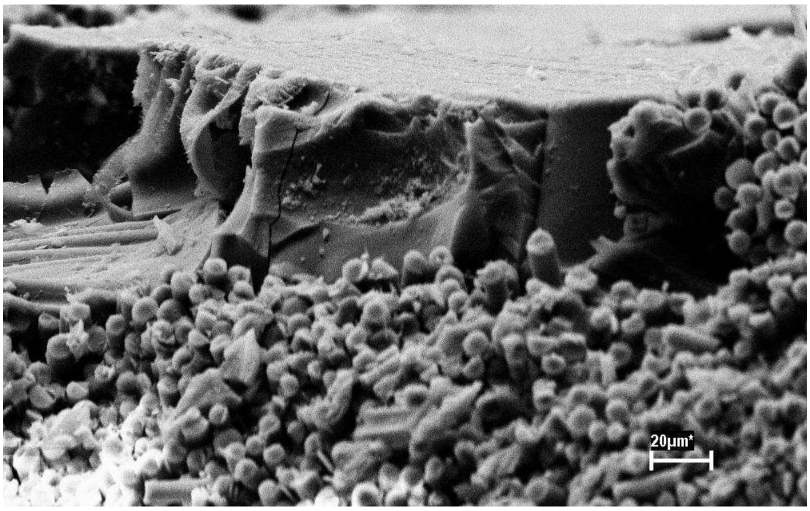


Figure 13. Surface of the quartz phenolic composite after the plasma torch burning test

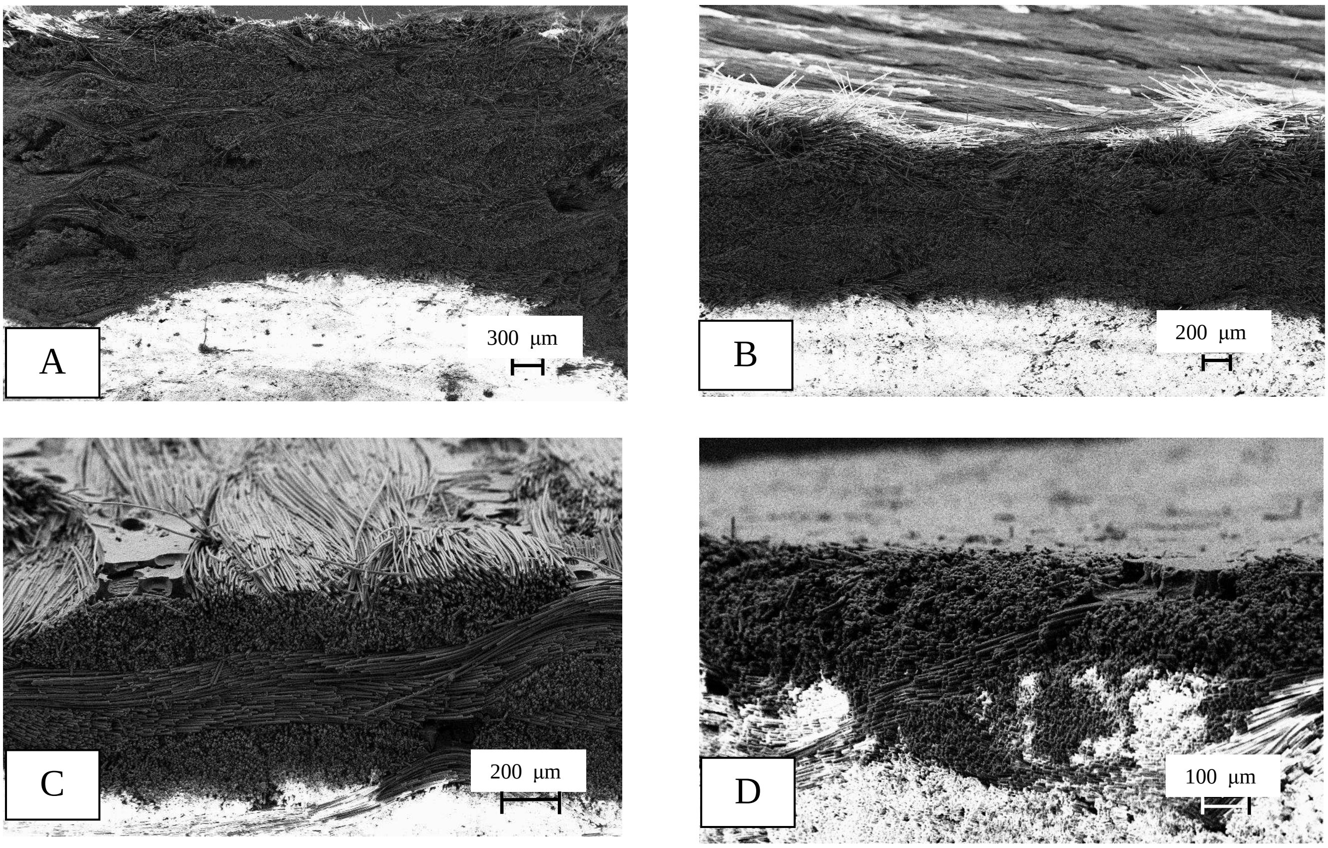


Figure 14. Side view of the specimens submitted to the plasma torch burning test

6. CONCLUSIONS

Quartz phenolic composites were prepared and evaluated by the measurement of the mechanical and ablative properties, in order to have the range of properties suitable for the use of these materials as thermal protection systems.

The mechanical properties evaluated were tensile, shear and flexural strength. The results obtained showed that this material has average values of 38,5 MPa, 52 MPa and 85 MPa for tensile, shear and flexural strength, respectively.

As-cured and post-cured samples were analyzed for the Iosipescu shear strength and it was found that the post-cured samples shows a shear strength of 19 ± 2 MPa while for the post-cured samples the value is 52 ± 2 MPa.

The as-cured quartz phenolic composite has a higher deformation up to failure (~30000 μm/m), in comparison with the post-cured one (~4500 μm/m). This means that post-cure is beneficial for composite properties providing it is not overcured, which may lead to property degradation.

It was not possible to compare the data obtained with one published in literature because as quartz phenolic composites are used in sensitive areas of aerospace technology the data are not easily available in literature.

The ablative tests show that the mass loss per unit area depends strongly on the temperature of the material surface and on the distance between the nozzle tip of the plasma gun and the front surface of the specimen

The information obtained from the plasma test indicates that this composite has ablation resistance and is reliable for the construction of thermal protection systems.

7. REFERENCES

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